

Adaptive fuzzy sliding mode control for gantry crane as varying rope length

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Abstract: Gantry crane is used quite commonly in hazardous areas, which increasingly requires strict control of the gantry crane operation process to improve efficiency and ensure safe gantry crane operation. Automated the gantry crane operating process is being applied popular currently. Gantry crane is often affected by large noise, having the varying- model parameters, so that proposed a adaptive fuzzy combining sliding mode controller for the gantry crane in this article. This control method derived from combining the sliding surfaces of three subsystem of the gantry crane (trolley position, rope length, anti-swing) to draw out two system sliding surfaces: the trolley position with the anti-swing and the rope length and the anti-swing. On the based of the sliding mode control principle, drawn out the equivalent controller and the switching controller for gantry crane. But due to the uncertain parameters - nonlinear model of gantry crane with the bound disturbances, combining the fuzzy approximate method, defined the fuzzy controller (used to mimic the equivalent controller) and the compensation controller for the difference between the equivalent controller and the fuzzy controller (used as the switching controller) for two system control inputs: trolley position and rope length. The adaptive control laws for these controllers were deduced from Lyapunov's stable criteria to asymptotically stabilize the sliding surfaces. Simulation results demonstrated the feasibility of the suggested method through gantry crane in the hazard areas.

Keyword: gantry crane, adaptive controller, adaptive fuzzy controller, sliding mode control, varying rope length

I. INTRODUCTION

In recent years, many modern gantry cranes have brought to the increasingly common use in the transport industry for loading and unloading of goods, in the construction industry to move materials or in the manufacturing industry to assembly of heavy equipments that can not use human power. According to the traditional operation way, as moving the trolley, the hoisting rope length is fixed. However in some cases, to reduce the transportation process time, we can operate simultaneously the gantry crane moving the trolley and lifting/lowering the payloads. This will increase the risk of unsafety for people and equipments because of oscillating of the package. Although following on experimented operators, can reduce the oscillation of the payload, but that would add reducing the efficiency of freight. Therefore, automating the gantry crane operation process to enhance accuracy simultaneously for moving the trolley and lifting/lowering payload, anti-swing angle of payload, reducing transportation time, increasing safety during crane operation, is essential and is also the most basic requirements of the gantry crane operation.

Control problem of the gantry crane is always attractive in many science researchers and have been published on many papers in prestige science journals. Allmost of controllers of the gantry crane are designed based on the nonlinear model or the linearized model, and unchangeable rope length. In [1] presented an adaptive controller for a linearized crane system. In [2] considered a linearized parameter-varying model of a planar crane and proposed an observer-based control design via Lyapunov equivalence function. Paper [3] dealt with adaptive sliding mode fuzzy control approach for a linearized two dimension overhead crane system. In [4,5,6] presented PID controller combined with fuzzy logic for a linearized crane model. These methods based on the linearized crane model may lose the sufficient accuracy of information about position and load swing, so that some uncertain factors may reduce the performance of these crane control systems.

Having some papers dealt with the crane control design based on nonlinear dynamical models. In [7] presented the adaptive fuzzy controller based on the trolley position error and sway angle error. The article [8] referred to the nonlinear optimal control. The article [9] presented the neural network controller for the crane. The sliding mode control (SMC) method can be found in the articles as follows: the coupling SMC [10,11], the SMC combined with fuzzy logic [12], the adaptive SMC [13, 14], the decoupling SMC [15]. The SMC method was proceeded based on the characteristics of the crane: the trolley and the hoist lifting/lowering the payload as unchanging rope length. For instance, [10,11] defined a sliding surface coupling both subsystems, [12] split a crane system and rebuilt the system states to achieve the hierarchical SMC controllers, [13] utilized such the divisibility to design a fuzzy sliding surface for trolley subsystem. The divisibility of crane systems were also adopted for the control design in [14,15].

Recently, several researches focused towards improving efficiency of crane operation, rapidly quenchinh the payload swing, despite of the trolley movement and lifting/lowering the winch. In [16,17] presented the anti-swing control of crane with changing rope length, in which the trolley motion rapidly achieved the suppression of the load sway and the asymptotic stability of the zero solutions of the position and rope length tracking errors was proved. In [18,19] dealt with the controllers having compensated friction but for simply crane model, not dealing with the bound disturbance hazard areas such as wind velocity, dynamic friction.

In this paper, the gantry crane - three degree of freedom is investigated (trolley position, rope length, and sway angle), but the number of actutors is only two (trolley and hoist motors). Based on Euler-Lagrange's energy ballance equation, a dynamic model of the gantry crane is presented in section 2. In section 3, defined sliding surfaces for three decoupling subsystem of the gantry crane and then combining each pair system sliding surfaces: trolley with sway angle; and rope length with sway angle. Based on the combining sliding surfaces, proposed a new adaptive fuzzy combining sliding mode control for the gantry crane as varying the rope length and impacted by the bound disturbance hazard areas. Here proved this proposed controller according to the asymptotic stability criteria of Lyapunov's function. Then validity of the proposed method in section 4 is illustrated through simulation results. Finally, conclusions are given in section 5.

II. GANTRY DYNAMIC MODEL

The gantry crane is described in Fig 1. Movement of gantry crane is procedured by the trolley and hoist motors, corresponding to two control force F_x and F_l . Position of the load is given in $2D - Oxz$, caused by trolley motion and hoist motion, $x(t)$ - the displacement of the trolley in the x -direction, $l(t)$ - the hoist rope length in the l -direction, $\alpha(t)$ - the sway angle of the load on x - z suface, $F_f(x)$ - the friction force of the trolley, $F_n(x)$ - the bound disturbance force impacting into the trolley. We have the following assumptions: the payload and trolley are connected by a massless rigid rod throught the hoist; the trolley and the payload can be regarded as point masses; the rope elongation due to tension force is negligible.

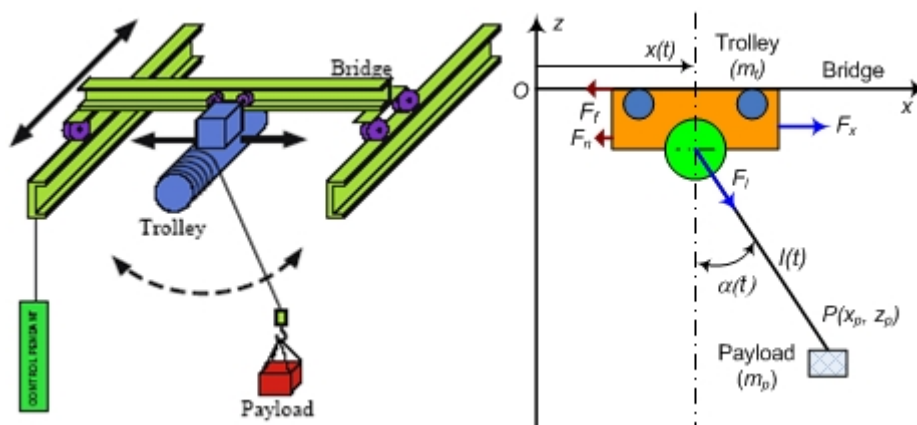


Fig 1. Constructure and diagram of the gantry crane

Let the coordinate of the payload be (x_p, z_p) , m_p is the payload mass, m_t is the trolley mass, g is the gravitational acceleration, then the system kinetic energy T and the system potential energy V are given as follows:

$$T = T_{trolley} + T_{load} = \frac{1}{2} m_t \dot{x}^2 + \frac{1}{2} m_p (\dot{x}_p^2 + \dot{z}_p^2); \quad x_p = x + l \sin \alpha, \quad z_p = -l \cos \alpha \quad (1)$$

$$V = V_{trolley} + V_{load} = -m_p g l \cos \alpha$$

Constructing the system Lagrange $L=T-V$, we can obtain the following Lagrange equation related to the generalized coordinates $q = [x, l, \alpha]^T$ as:

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}}(q, \dot{q}) \right) - \frac{\partial L}{\partial q}(q, \dot{q}) = \tau; \quad \tau = [F_x - F_f - F_n, F_l, 0]^T \quad (2)$$

Finally, the equations of motion using the Lagrange's equation are derived as follows:

$$\begin{cases} (m_t + m_p)\ddot{x} + m_p(\sin \alpha)\ddot{l} + m_p l(\cos \alpha)\ddot{\alpha} + 2m_p(\cos \alpha)\dot{l}\dot{\alpha} - m_p l(\sin \alpha)\dot{\alpha}^2 = F_x - F_f - F_n \\ m_p(\sin \alpha)\ddot{x} + m_p\ddot{l} - m_p l\dot{\alpha}^2 - m_p g \cos \alpha = F_l \\ (\cos \alpha)\ddot{x} + \ddot{l} + 2\dot{l}\dot{\alpha} + g \sin \alpha = 0 \end{cases} \quad (3)$$

Suppose that total of the friction and disturbance force $(-F_f - F_n)$ is expressed as $d(x)$, in which $|d(x)| \leq D$, the uncertainty bound D is a positive constant, control input $u_1 = F_x, u_2 = F_l$, state variables $x_1 = x(t), x_2 = \dot{x}, x_3 = l(t), x_4 = \dot{l}, x_5 = \alpha(t), x_6 = \dot{\alpha}$ then above dynamic model can be transformed to the state space expression as follows:

$$\begin{cases} \dot{x}_1 = x_2 \\ \dot{x}_2 = f_1 + g_1 u_1 + h_1 u_2 \\ \dot{x}_3 = x_4 \\ \dot{x}_4 = f_2 + g_2 u_1 + h_2 u_2 \\ \dot{x}_5 = x_6 \\ \dot{x}_6 = f_3 + g_3 u_1 + h_3 u_2 \end{cases} \quad \text{with} \quad \begin{cases} f_1 = d / m_t, g_1 = 1 / m_t, h_1 = -(\sin x_3) / m_t \\ f_2 = x_3 x_6^2 + g \cos x_5, g_2 = -(\sin x_5) / m_t, \\ h_2 = 1 / m_p + (\sin^2 x_5) / m_t \\ f_3 = -(g \sin x_5 + 2x_4 x_6) / x_3, \\ g_3 = -(\cos x_5) / (m_t x_3), \\ h_3 = (\cos x_5 \sin x_5) / (m_t x_3) \end{cases} \quad (4)$$

Thus, the gantry crane system is divided into three coupled subsystems: the positioning subsystem (x_1, x_2) , rope length subsystem (x_3, x_4) and anti-swing subsystem (x_5, x_6) . The control object of the gantry crane is to move the trolley and the varying rope length to the destination and almost anti-swing of the payload when the system model exists uncertainty parameters and disturbances (for example, winds and different payloads).

III. CONTROL DESIGN

A. Sliding mode control for gantry crane

Assume that $x_d = x_{1d}, l_d = x_{3d}, x_{5d} = \alpha_d$ are the input reference trajectory of trolley position, rope length and swing angle, respectively, in generally, $\alpha_d = 0$. Define the tracking error as: $e_1 = x_1 - x_d, e_3 = x_3 - l_d, e_5 = x_5 - \alpha_d$ and three sliding mode functions for three subsystems as follows: $s_x = \dot{e}_1 + c_1 e_1, s_l = \dot{e}_3 + c_3 e_3, s_\alpha = \dot{e}_5 + c_5 e_5$, where c_1, c_3, c_5 are positive numbers.

In this article, a novel SMC method is applied here, by combining the sliding surfaces, i.e. using positive constants λ_1 and $\lambda_2 = 0$ to define a suitable pair of the system sliding surfaces as:

$$s = \begin{bmatrix} s_1 \\ s_2 \end{bmatrix} = \begin{bmatrix} s_x + \lambda_1 s_\alpha \\ s_l + \lambda_2 s_\alpha \end{bmatrix} = \begin{bmatrix} s_x + \lambda_1 s_\alpha \\ s_l \end{bmatrix} \quad (5)$$

Following on the methodology of equivalent control of variable structure control, the SMC law includes two parts: switching control law and equivalent control law. The switching control law is employed to drive the system states moving towards a specific sliding surface. The equivalent control law guarantees the system states to keep sliding on the sliding surface and converge to zero along the sliding surface. To design the combining SMC, we still adopt such the approach and define the total control law u as follows:

$$u = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} u_{eq1} + u_{sw1} \\ u_{eq2} + u_{sw2} \end{bmatrix} \quad (6)$$

where u_{eqi} is the equivalent control and u_{swi} is the switching control $u_i, (i = 1, 2)$

Letting $\dot{s} = 0$, and substituting (4) into it, the equivalent control on the surface s is gotten as:

$$u_{eq1} = \frac{-h_2 f_1 + (h_1 + \lambda_1 h_3) f_2 - \lambda_1 h_2 f_3 - c_1 h_2 x_2 + c_3 (h_1 + \lambda_1 h_3) x_4 - \lambda_1 c_5 h_2 x_2 + h_2 (\ddot{x}_d + c_1 \dot{x}_d) - (h_1 + \lambda_1 h_3) (\ddot{l}_d + c_3 \dot{l}_d)}{g_1 h_2 + \lambda_1 g_3 h_2 - g_2 h_1 - \lambda_1 g_2 h_3} \quad (7)$$

$$u_{eq2} = \frac{-f_2 - c_3 x_4 + \ddot{l}_d + c_3 \dot{l}_d - g_2 u_{eq1}}{h_2}$$

The equivalent control law u_{eq1} and u_{eq2} are complicated function, moreover function f_i is uncertain, thus, the following adaptive fuzzy combining sliding mode control is proposed here.

B. Adaptive fuzzy sliding mode control

Consider two fuzzy control system $j=1,2$ for the control input u_{fj} with the i -th fuzzy rule as:

$$\text{Rule } i: \text{ IF } s_j \text{ is } F_{ji} \text{ THEN } u_{fj} \text{ is } \theta_{ji}, i=1,2,\dots,n \tag{8}$$

where F_{ji} are the label of the fuzzy sets characterized by the fuzzy membership functions $\mu_{F_{ji}}(\cdot)$ and θ_{ji} are the adjustable fuzzy singletons.

The defuzzification of the output is accomplished by the method of center of gravity:

$$u_{fj}(s_j, \theta_j) = \frac{\sum_{i=0}^n \omega_{ji} \theta_{ji}}{\sum_{i=0}^n \omega_{ji}} = \theta_j^T \omega_j \tag{9}$$

where $\omega_{ji} = \mu_{F_{ji}}(s_j)$ is the firing weight of the i -th fuzzy rule;

$$\theta_j = [\theta_{j1}, \theta_{j2}, \dots, \theta_{jn}]^T; \omega_j = [\omega_{j1} / \sum_{i=1}^n \omega_{ji}, \omega_{j2} / \sum_{i=1}^n \omega_{ji}, \dots, \omega_{jn} / \sum_{i=1}^n \omega_{ji}]^T$$

To estimate optimal parameters of the fuzzy system (9), use $\hat{\theta}_j$ instead of θ_j , then:

$$u_{fj}(s_j, \hat{\theta}_j) = \hat{\theta}_j^T \omega_j \tag{10}$$

According to fuzzy approximated theory, there exists an optimal fuzzy system $u_{fj}^*(t)$ such that

$$u_{fj}^*(t) = u_{fj}^*(s_j, \theta_j^*) = \theta_j^{*T} \omega_j \tag{11}$$

where the time invariant optimal parameter vector θ_j^* is defined as

$$\theta_j^* = \arg \min_{|\theta_j| \leq M_{\theta_j}} \left\{ \sup_{|s_j| \leq M_{s_j}} |u_{fj}(s_j, \theta_j) - u_{eqj}| \right\}, \forall t \tag{12}$$

and M_{θ_j}, M_{s_j} are specified by the designer. The minimum approximation error is defined as

$$\gamma_j(t) = u_{fj}^* - u_{eqj} \quad 0 \leq |\gamma_j(t)| \leq \Gamma_j \tag{13}$$

where the uncertainty bound $\hat{\Gamma}_j$ is a positive constant. However, this uncertainty bound can not be measured for practical application. Thus, a bound estimation $\hat{\Gamma}_j$ will be developed to estimate the approximation error bound Γ_j of the minimum approximation error $\gamma_j(t)$. Define the estimation error $\tilde{\Gamma}_j = \Gamma_j - \hat{\Gamma}_j$, then the system control law is assumed to take the form as:

$$u_j(s_j, \hat{\theta}_j, \hat{\Gamma}_j) = u_{fj}(s_j, \hat{\theta}_j) + u_{cpj}(s_j, \hat{\Gamma}_j) \tag{14}$$

where the fuzzy controller u_{fj} given in (9) is used to mimic the equivalent controller in (6); and the compensation controller u_{cpj} given in the following is used as the switching controller in (6).

Following, we identify the compensation controller u_{cpj} . Define a Lyapunov function as:

$$V(s, \tilde{\theta}, \tilde{\Gamma}) = \frac{1}{2} s^T s + \frac{1}{2\eta_1} \tilde{\theta}^T \tilde{\theta} + \frac{1}{2\eta_2} \tilde{\Gamma}^T \tilde{\Gamma} \tag{15}$$

where $\tilde{\theta} = \begin{bmatrix} \tilde{\theta}_1 \\ \tilde{\theta}_2 \end{bmatrix} = \begin{bmatrix} \theta_1^* - \hat{\theta}_1 \\ \theta_2^* - \hat{\theta}_2 \end{bmatrix}; \tilde{\Gamma} = \begin{bmatrix} \tilde{\Gamma}_1 \\ \tilde{\Gamma}_2 \end{bmatrix} = \begin{bmatrix} \Gamma_1 - \hat{\Gamma}_1 \\ \Gamma_2 - \hat{\Gamma}_2 \end{bmatrix}$

Differentiate V with respect to time t , derive as:

$$\dot{V}(s, \tilde{\theta}, \tilde{\Gamma}) = s_1 \dot{s}_1 + s_2 \dot{s}_2 + \frac{1}{\eta_1} (\tilde{\theta}_1^T \dot{\tilde{\theta}}_1 + \tilde{\theta}_2^T \dot{\tilde{\theta}}_2) + \frac{1}{\eta_2} (\tilde{\Gamma}_1 \dot{\tilde{\Gamma}}_1 + \tilde{\Gamma}_2 \dot{\tilde{\Gamma}}_2) \quad (16)$$

From (5) and (13-14), derived as:

$$\begin{aligned} \dot{s}_1 &= f_1 + \lambda_1 f_3 + c_1 x_2 + \lambda_1 c_2 x_6 - \ddot{x}_d - c_1 \dot{x}_d + (g_1 + \lambda_1 g_3)(u_{f1} + u_{cp1} - u_{f1}^* + u_{f1}^*) \\ &\quad + (h_1 + \lambda_1 h_3)(u_{f2} + u_{cp2} - u_{f2}^* + u_{f2}^*) \\ \dot{s}_1 &= f_1 + \lambda_1 f_3 + c_1 x_2 + \lambda_1 c_2 x_6 - \ddot{x}_d - c_1 \dot{x}_d + (g_1 + \lambda_1 g_3)(u_{f1} + u_{cp1} - u_{f1}^* + \gamma_1 + u_{eq1}) \\ &\quad + (h_1 + \lambda_1 h_3)(u_{f2} + u_{cp2} - u_{f2}^* + \gamma_2 + u_{eq2}) \\ \dot{s}_1 &= (g_1 + \lambda_1 g_3)(u_{f1} - u_{f1}^* + u_{cp1} + \gamma_1) + (h_1 + \lambda_1 h_3)(u_{f2} - u_{f2}^* + u_{cp2} + \gamma_2) \\ \dot{s}_2 &= f_2 + \lambda_2 x_4 - \ddot{l}_d - \lambda_2 \dot{l}_d + g_2(u_{f1} + u_{cp1} - u_{f1}^* + u_{f1}^*) + h_2(u_{f2} + u_{cp2} - u_{f2}^* + u_{f2}^*) \\ \dot{s}_2 &= f_2 + \lambda_2 x_4 - \ddot{l}_d - \lambda_2 \dot{l}_d + g_2(u_{f1} + u_{cp1} - u_{f1}^* + \gamma_1 + u_{eq1}) + h_2(u_{f2} + u_{cp2} - u_{f2}^* + \gamma_2 + u_{eq2}) \\ \dot{s}_2 &= g_2(u_{f1} - u_{f1}^* + u_{cp1} + \gamma_1) + h_2(u_{f2} - u_{f2}^* + u_{cp2} + \gamma_2) \end{aligned} \quad (17)$$

And then, combined (16) and (17), as putting $g_{13} = g_1 + \lambda_1 g_3$, $h_{13} = h_1 + \lambda_1 h_3$, derive as:

$$\begin{aligned} \dot{V} &= s_1 [g_{13}(u_{f1} - u_{f1}^* + u_{cp1} + \gamma_1) + h_{13}(u_{f2} - u_{f2}^* + u_{cp2} + \gamma_2)] + \\ &\quad + s_2 [g_2(u_{f1} - u_{f1}^* + u_{cp1} + \gamma_1) + h_2(u_{f2} - u_{f2}^* + u_{cp2} + \gamma_2)] + \frac{1}{\eta_1} (\tilde{\theta}_1^T \dot{\tilde{\theta}}_1 + \tilde{\theta}_2^T \dot{\tilde{\theta}}_2) + \frac{1}{\eta_2} (\tilde{\Gamma}_1 \dot{\tilde{\Gamma}}_1 + \tilde{\Gamma}_2 \dot{\tilde{\Gamma}}_2) \\ \dot{V} &= s_1 [g_{13}(-\tilde{\theta}_1^T \omega_1 + u_{cp1} + \gamma_1) + h_{13}(-\tilde{\theta}_2^T \omega_2 + u_{cp2} + \gamma_2)] + \\ &\quad + s_2 [g_2(-\tilde{\theta}_1^T \omega_1 + u_{cp1} + \gamma_1) + h_2(-\tilde{\theta}_2^T \omega_2 + u_{cp2} + \gamma_2)] + \\ &\quad + \frac{1}{\eta_1} (\tilde{\theta}_1^T \dot{\tilde{\theta}}_1 + \tilde{\theta}_2^T \dot{\tilde{\theta}}_2) + \frac{1}{\eta_2} (\Gamma_1 - \hat{\Gamma}_1)(-\dot{\hat{\Gamma}}_1) + \frac{1}{\eta_2} (\Gamma_2 - \hat{\Gamma}_2)(-\dot{\hat{\Gamma}}_2) \\ \dot{V} &= \frac{1}{\eta_1} \tilde{\theta}_1^T [\dot{\tilde{\theta}}_1 - \eta_1 \omega_1 (g_{13} s_1 + g_2 s_2)] + \frac{1}{\eta_1} \tilde{\theta}_2^T [\dot{\tilde{\theta}}_2 - \eta_1 \omega_2 (h_{13} s_1 + h_2 s_2)] + \\ &\quad + (g_{13} s_1 + g_2 s_2) u_{cp1} + \frac{1}{\eta_2} \hat{\Gamma}_1 \dot{\hat{\Gamma}}_1 + (g_{13} s_1 + g_2 s_2) \gamma_1 - \frac{1}{\eta_2} \Gamma_1 \dot{\hat{\Gamma}}_1 \\ &\quad + (h_{13} s_1 + h_2 s_2) u_{cp2} + \frac{1}{\eta_2} \hat{\Gamma}_2 \dot{\hat{\Gamma}}_2 + (h_{13} s_1 + h_2 s_2) \gamma_2 - \frac{1}{\eta_2} \Gamma_2 \dot{\hat{\Gamma}}_2 \end{aligned} \quad (18)$$

According to Lyapunov's stability criteria, the necessary condition for the control system stability: $\dot{V} \leq 0$. Thereby, the compensation controller u_{cpj} is given in (19) with the bound estimation law presented in (20), moreover can derive the adaptive law $\hat{\theta}_j$ given in (21).

$$u_{cp1} = -\hat{\Gamma}_1 \operatorname{sgn}(g_{13} s_1 + g_2 s_2); \quad u_{cp2} = -\hat{\Gamma}_2 \operatorname{sgn}(h_{13} s_1 + h_2 s_2) \quad (19)$$

$$\dot{\hat{\Gamma}}_1 = -\dot{\tilde{\Gamma}}_1 = \eta_2 |g_{13} s_1 + g_2 s_2|; \quad \dot{\hat{\Gamma}}_2 = -\dot{\tilde{\Gamma}}_2 = \eta_2 |h_{13} s_1 + h_2 s_2| \quad (20)$$

$$\dot{\hat{\theta}}_1 = -\dot{\tilde{\theta}}_1 = -\eta_1 \omega_1 (g_{13} s_1 + g_2 s_2); \quad \dot{\hat{\theta}}_2 = -\dot{\tilde{\theta}}_2 = -\eta_1 \omega_2 (h_{13} s_1 + h_2 s_2) \quad (21)$$

Substituting (20-22) into (19), obtained as:

$$\dot{V} = (g_{13} s_1 + g_2 s_2) \gamma_1 - |g_{13} s_1 + g_2 s_2| \Gamma_1 + (h_{13} s_1 + h_2 s_2) \gamma_2 - |h_{13} s_1 + h_2 s_2| \Gamma_2 \leq 0 \quad (22)$$

The negative semidefiniteness of the Lyapunov function guarantees that $s, \tilde{\theta}, \tilde{\Gamma}$ are bounded, i.e.

function V approaches to zero ($V \rightarrow 0$) as time approaches to infinite ($t \rightarrow \infty$), guaranteed that:

$$\begin{aligned} s \rightarrow 0 &\Rightarrow s_1 \rightarrow 0, s_2 \rightarrow 0 \\ \tilde{\theta} \rightarrow 0 &\Rightarrow \hat{\theta}_1 \rightarrow \theta_1^*, \hat{\theta}_2 \rightarrow \theta_2^* \\ \tilde{\Gamma} \rightarrow 0 &\Rightarrow \hat{\Gamma}_1 \rightarrow \Gamma_1, \hat{\Gamma}_2 \rightarrow \Gamma_2 \end{aligned} \quad (23)$$

Hence, the nonlinear gantry crane system presented in (4), the control law is designed as in (14), in which the fuzzy controller u_{fzj} given in (9) with the adaptive law given in (21) and the compensation controller u_{cpj} is given in (19) with the bound estimation presented in (20), then the stability of the gantry crane system can be guaranteed.

IV. SIMULATION RESULT

In this section, we proceed the gantry crane system simulation with the suggested controller in Matlab to validate this work. It should be emphasized that the derivation of the proposed adaptive fuzzy combining sliding mode control (AFCSMC) does not need to use the dynamic function $f_i(x), g_i(x), h_i(x)$ in (4), $i=1,2,3$. These dynamic functions are used only for simulations. This is the best advantage of the proposed control method than a lot of nonlinear control methods [3-15].

The control objective of the gantry crane control is to transport the payload to the required destination as fast and as accurately as possible without swings. The physical parameters of the gantry crane system in Fig 1 are determined as follows: $g=9.8[m^2/s]$, $m_t=2[kg]$, $m_p=0.2[kg]$, $l=1[m]$, $x=1[m]$, $c_1=0.5, c_3=0.6, c_5=0.8$, $\lambda_1=0.7, \lambda_2=0, \eta_1=50, \eta_2=0.001$ and the membership functions are constructed for the antecedent part; the value of the center of the triangular type membership functions are given as $[-3.5 \ -1.5 \ 0.5 \ 0.5 \ 1.5 \ 3.5]$. These parameters are chosen through some trials.

The block diagram of the adaptive fuzzy combining sliding mode control for the gantry crane is described in Fig 2. The AFCSMC controller consists of the sliding surfaces block given in (5), the fuzzy controller block given in (9) with the adaptive law given in (21) and the compensation controller block given in (19) with the bound estimation presented in (20).

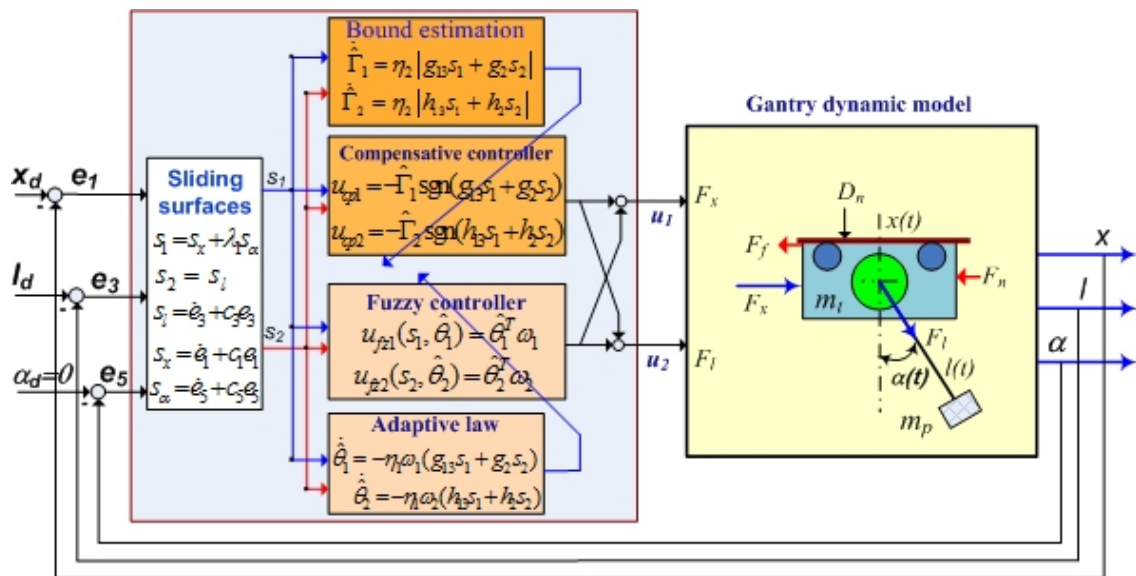


Fig 2. Proposed adaptive fuzzy combining sliding mode control for gantry crane

The simulation was carried out in three crane operating circumstances: (1) - changing the trolley position without rope length, (2) - varying the rope length without the trolley movement, (3) - changing both the trolley position and the rope length, and using the bound disturbance hazard areas (including friction force and disturbance force) described in Fig 3. Simulation results of the AFCSMC as comparing with the PID controller [19,20] were presented in Fig 3. The simulation script was as follow: the first time $t=0$, the trolley moved to the location $0.5m$ and lifting the payload at high $0.6m$; then $t=10sec$, varied the rope length to lower height $0.2m$; after finishing the additional payload process $t=20sec$, varied the rope length to lift up the payload at high $1m$ and, and at the same time drove the trolley to destination $0.9m$; before the trolley stopped at target, it should be varied the rope length to lower the payload, possibly at time $t=30sec$.

The simulation results in Fig 3 showed that as having the bound noise impacted in the gantry crane with small margin (as Fig 3), both the PID controller [19,20] and AFCSMC controller are to meet all requirements of the gantry crane operation, satisfied the payload position (trolley position, rope length) and the sway angle, but AFCSMC for better quality control: faster response time (trolley position, rope length), eliminated steady error, smaller payload oscillating amplitude ($<7^\circ$) and quenched rapidly oscillating payload ($<5sec$) despite of moving simultaneously the trolley and lifting/lowering the payload.

However, when a large noise (as Fig 3), the PID controller does not meet operational requirements of the gantry crane: the trolley position was strongly oscillated, the sway angle of payload was oscillated sine sharp or with ascending amplitude ($>90^\circ$), while the AFCSMC controller still allowed smoothly the gantry crane operation: responded the trolley position was small swing around the reference trajectory but still within tolerance limit ($<2\%$), at the departure time $t = 0.5sec$, swing angle was quite large oscillated (maximums amplitude $<15^\circ$), then fell rapidly down the small oscillation amplitude ($<9^\circ$). This proved that the proposed AFCSMC controller can be applied to operate the gantry crane in the hazardous areas.

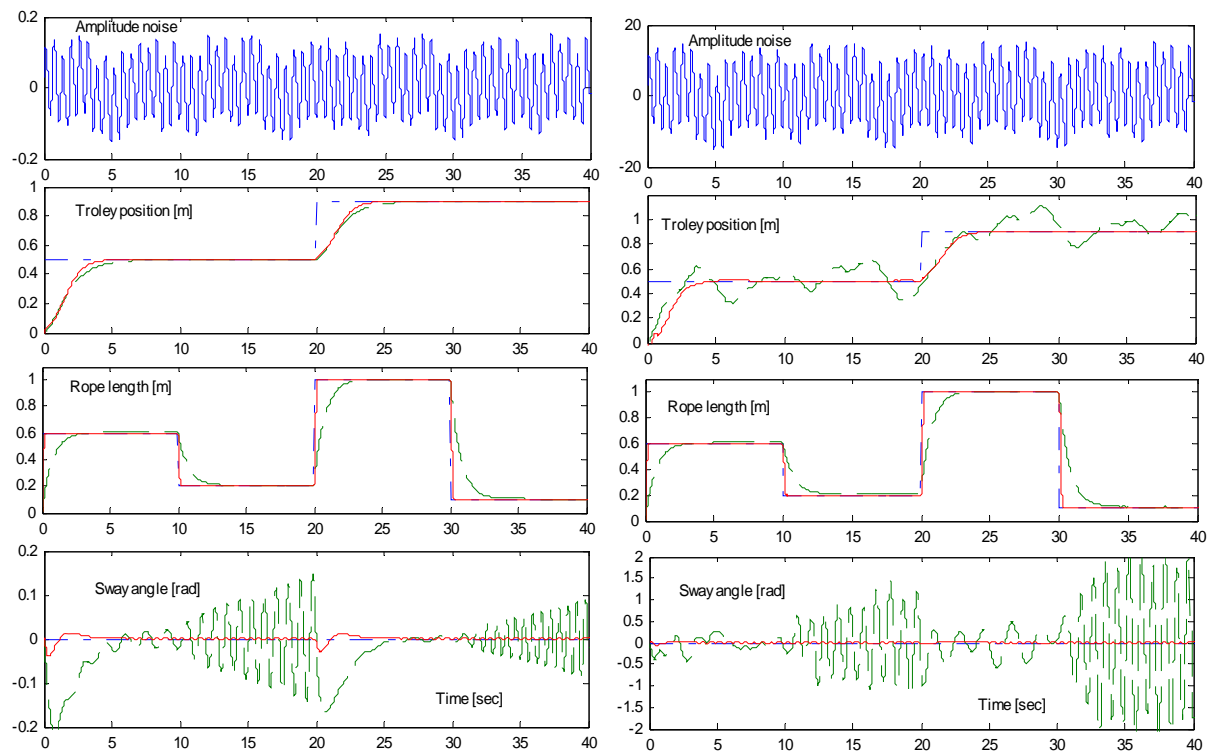


Fig 3. System performance with the AFCSMC and PID controllers, noted ‘-.-’ as reference trajectory, ‘-.-’ response curves with PID, ‘-.-’ response curves with AFCSMC.

V. CONCLUSION

Gantry crane is always operated in hazardous areas and influenced of large disturbances, so the gantry crane mathematical model is almost not defined. This article dealt with a gantry crane model, allow simultaneously to moving the trolley and the lifting/lowering payload, from that proposed a adaptive fuzzy sliding mode control on the base of combination of the system sliding surfaces: the trolley with the swing angle and the rope length with the swing angle. The AFCSMC controller was drawn and proven according to the stability criteria of Lyapunov’s function, guaranteed sufficient condition of the stability of the system states. Simulation results validated the proposed controller can applied into the transport control problem of the gantry crane system in the hazardous areas.

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